1.	(Currently Amended)	A hybrid blade (1)-for thermal turbomachines, having the
blade	comprising:	
	an airfoil (2) made of	a first metallic material of a certain first density, and having;
	_a blade root (3), chara	cterized in that the blade root (3), compared with the airfoil (2), is
made	of a different second me	etallic material having a lower second density lower than the first
<u>densi</u>	ty;; and in that	
	wherein the airfoil (20) is connected to the blade root (3) in a positive-locking manner.
2.	(Currently Amended)	The hybrid blade (1) as claimed in claim 1, eharacterized in
that s	aid wherein the blade (1) is comprises a compressor blade, in particular a high-pressure
comp	oressor blade, in which;	
	wherein the airfoil (2)	is made of a stainless CrNi steel; and
	wherein the blade root	(3) is made of a material selected from the group consisting of a
high-	temperature titanium all	oy, or an intermetallic gamma titanium aluminide alloy, or and an
interr	metallic orthorhombic tit	anium aluminide alloy.
3.	(Currently Amended)	The hybrid blade (1) as claimed in claim 1, characterized in
that s	wherein the blade (1) is <u>comprises</u> a turbine blade<u>; in which</u>
	wherein the airfoil (2)	is made of a superalloy, for example a nickel-based superalloy,;
and		
	wherein the blade roo	(3) is made of a material selected from the group consisting of a
high-	temperature titanium all	oy, or an intermetallic gamma titanium aluminide alloy, or and an
interr	metallic orthorhombic tit	anium aluminide alloy.
4.	(Currently Amended)	The hybrid blade (1) as claimed in one of claims 1 to
3clain	m 1, characterized in tha	t said wherein the blade (1) is comprises a moving blade.
5.	(New) The hy	brid blade as claimed in claim 2, wherein the compressor blade

comprises a high-pressure compressor blade.

6. (New) The hybrid blade as claimed in claim 3, wherein the superalloy comprises a nickel-based superalloy.